

CHANDRAYAAN-1

INDIA'S FIRST MISSION TO MOON

Expanding scientific knowledge about the moon, upgrading India's technological capability and providing challenging opportunities for planetary research to the younger generation

Spacecraft:

Physical : 1.5 m cuboid-shaped main body with a mass of about 600 kg in 100 km lunar orbit

Control : 3-axis stabilised using reaction wheels and attitude control thrusters; sun sensors, star sensors, and miniaturised DTGs for

Power : Single sided solar array to generate 700 W, 36Ah Li-Ion battery for eclipse

orientation reference

operations

Propulsion : Bi-propellant system for transferring
from Initial (Earth) Orbit to Lunar Orbit and

for orbit and orientation maintenance

<u>Communication</u>: TTC in S-band and scientific payload

data transmission in X-band
Mission Life : 2 years

Mission Profile:

- Launched on October 22, 2008 into 250 km x 23,000 km Initial (Earth) Orbit by PSLV-C11
- Solar panel deployment
- Multiple perigee manoeuvres to achieve 380,000 km apogee [Lunar Transfer Trajectory (LTT)]
 17 days in Earth Bound Orbits prior to lunar encounter

Earth and moon imaging by TMC and RADOM Switch ON

- Lunar capture and Spacecraft in 504 km x 7502 km orbit of the moon
- Orbit Manoeuvres to reduce lunar orbit height to 100 km circular polar
- MIP separation and impact
- Payloads switch ON followed by beginning of normal phase

Scientific Objective:

High resolution remote sensing of moon in the visible, near infrared, microwave, low and high energy X-ray regions for:

- Preparing 3-dimensional atlas of the lunar surface
- Chemical mapping of entire lunar surface

Indian Scientific Payloads:

- Terrain Mapping Camera (TMC) with stereo imaging capability and 5 m spatial resolution
- Hyper-Spectral Imager (HySI) operating in 0.4-0.95 micrometre band with a spectral resolution of 15 nanometre, spatial resolution of 80 m
- Lunar Laser Ranging Instrument (LLRI) with a height resolution of 5 m
- High Energy X-ray Spectrometer (HEX) operating in 30-270 kilo electron Volts (keV)
- Moon Impact Probe (MIP) to prove technology elements for future landing missions

ESA Payloads:

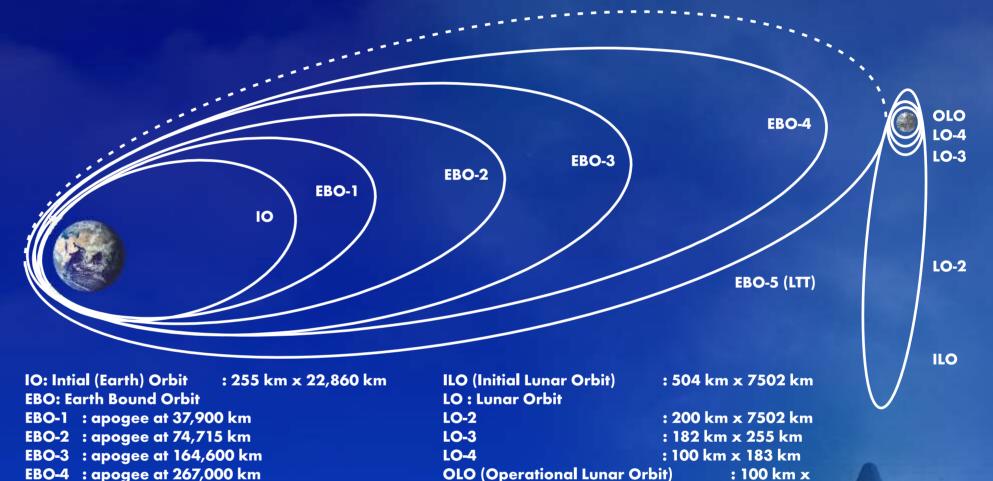
- Chandrayaan-1 Imaging X-ray Spectrometer (C1XS) from Rutherford Appleton Laboratory, UK and ISRO Satellite Centre operating in1-10 keV
- Smart near Infrared Spectrometer (SIR-2) from Max-Planck Institute of Aeronomy, Germany operating in 0.9-2.4 micrometre band
- Sub keV Atom Reflecting Analyser of Swedish Institute of Space Physics (SARA), with contribution from Vikram Sarabhai Space Centre

US Payloads (Through NASA):

- Miniature Synthetic Aperture Radar (Mini-SAR) from APL, Johns Hopkins University and Naval Air warfare Centre
- Moon Mineralogy Mapper (M3) operating in 0.7-3 micrometre band from Brown University and JPL, NASA

Bulgarian Payload:

 Radiation Dose Monitor (RADOM) from Bulgarian Academy of Sciences



Polar Satellite Launch Vehicle (PSLV-C11)

Lift-off weight : 320 tonneHeight : 44.4 m

EBO-5 (LTT-Lunar Transfer Trajectory): apogee at

- O 13 consecutive successes of PSLV by October 2008
- O Demonstrated multiple satellite launch capability
- O Launched Satellites to Sun Synchronous, Geosynchronous Transfer, Low Earth and Highly Elliptical Earth Orbits

Payload capability:

- 600 kg into 100 km lunar orbit
- 1380 kg into initial earth Orbit (250 km x 23,000 km)

New Facilities:

- Indian Deep Space Network located near Bangalore with 18 m and 32 m diameter X/S-band antennas
- Indian Space Science Data Centre to process and archive scientific data
- Spacecraft Control Centre at ISRO Telemetry, Tracking and Command Network







Lift-off of PSLV-C11carrying Chandrayaan-1 from the Second Launch Pad at Satish Dhawan Space Centre SHAR, Sriharikota

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